## Vibrations of an Impeller

## Introduction

This tutorial model demonstrates the use of dynamic cyclic symmetry with postprocessing on the full geometry. A 3D impeller with eight identical blades can be divided into eight sectors of symmetry. The model computes the fundamental frequencies for the full impeller geometry and compares them to the values computed for a single sector with the Floquet periodicity boundary conditions applied on two sector boundaries. It also demonstrates how to set up a frequency response analysis for one sector of symmetry, and how to postprocess the results into the full geometry by using the general extrusion coupling variables. The results for one sector are in very good agreement with the computations on the full geometry, while both the computational time and memory requirements are significantly reduced.

## Model Definition

Figure 1 shows the impeller geometry. The problem is solved using the cartesian coordinate system in 3D.


Figure 1: Impeller geometry.

The geometry can be divided into eight identical parts, each represented by a sector with an angle $\theta=\pi / 4$ about the $z$-axis; see Figure 2 .


Figure 2: Sector of periodicity.
The impeller is made of aluminum, and is supposed to be mounted on a shaft. The mounting boundary is modeled via a fixed constraint, and all possible effects of the shaft rotation are neglected.

The analysis is based on the Floquet theory which can be applied to the problem of small-amplitude vibrations of spatially periodic structures, Ref. 1. This includes the case of cyclic symmetry studied in this model.

If the problem is to determine the frequency response to a small-amplitude time-periodic excitation that also possesses cyclic symmetry, the theory states that the solution can be sought in form of a product of two functions. One follows the periodicity of the structure, while the other one follows the periodicity of the excitation. The problem can be solved in one sector of periodicity by applying the corresponding periodicity conditions to each of the two components in the product. The situation is often referred to as dynamic cyclic symmetry.

For an eigenfrequency study, one can show that all the eigenmodes of the full problem can be found by performing the analysis on one sector of symmetry only and imposing the cyclic symmetry of the eigenmodes with an angle of periodicity $\varphi=m \theta$, where
the cyclic symmetry mode number $m$ can vary from 0 to $N / 2$, with $N$ being the total number of sectors so that $\theta=2 \pi / N$.

You model the problem using the full solution without applying the above described multiplicative decomposition. For such a solution, the Floquet periodicity conditions at the sides of the sector of symmetry can be expressed as

$$
\begin{equation*}
\mathbf{u}_{\text {destination }}=e^{-j \varphi} R_{\theta} \mathbf{u}_{\text {source }}^{T} \tag{1}
\end{equation*}
$$

where the $\mathbf{u}$ represents the displacement vectors with the components given in the default cartesian coordinates. Multiplication by the rotation matrix $R$ given by

$$
R_{\theta}=\left[\begin{array}{ccc}
\cos (\theta) & -\sin (\theta) & 0 \\
\sin (\theta) & \cos (\theta) & 0 \\
0 & 0 & 1
\end{array}\right]
$$

makes the corresponding displacement components in the cylindrical coordinate system differ by the factor $\exp (-j \varphi)$ only. The angle $\varphi$ represents either the periodicity of the eigenmode for an eigenfrequency analysis or the periodicity of the excitation signal in case of a frequency-response analysis.

## Results and Discussion

In the first part of the analysis, you perform an eigenfrequency analysis of a single sector of periodicity, and then of the full geometry. A sweep over all required values of the cyclic symmetry parameter recovers all the eigenfrequencies of the full model with decent accuracy. See the Modeling Instructions section for in-detail comparison of the results and discussion of the performance gains.

In the second part, you perform a frequency-response analysis. Again, first of the sector of periodicity, and then of the full impeller geometry. The excitation is a pressure load applied to all free boundaries of the impeller. You enter it as a normal component of the boundary load using the expression

$$
F_{n}=-p_{0} \exp [-j m \operatorname{atan}(Y / X)]
$$

using the magnitude of $p_{0}=10^{4} \mathrm{~Pa}$ and cyclic symmetry parameter $m=3$. The excitation frequency is 200 Hz . Figure 3 and Figure 4 show very good agreement between the results computed on the full and reduced geometry.


Figure 3: Frequency response computed on the sector of periodicity only, and then visualized over the full geometry.

Surface: Total displacement (m)


Figure 4: Frequency response computed for the full geometry.

## Notes About the COMSOL Implementation

## MESHING

You use an unstructured mesh with the same size of the mesh elements for both calculations on one sector of symmetry and on the full geometry, see Figure 5. This helps to compare the results for this tutorial model. In practice, the mesh used for computations on the sector could be much finer, so that the results obtained via such geometry reduction would provide significantly better resolution of the results under the same memory requirements as for the full geometry (with a coarser mesh).


Figure 5: Meshed geometry.

## FLOQUET PERIODICITY CONDITION

You set up the linear extrusion coupling variables to connect the boundaries of the sector, which is needed to set up the Floquet periodicity conditions Equation 1.

Note that when entering the constraint expression, you also need to manually enter the constraint force expression i to ensure the Hermitian symmetry of the constraint force Jacobian matrix in the discretized model. See the Modeling Instructions section for more details.

## POSTPROCESSING

You use the general extrusion coupling variables to propagate the results computed for one sector over the full geometry. The coordinate mapping is done as

$$
\mathbf{r}_{\text {destination }}=R_{s} \mathbf{r}_{\text {source }}^{T}
$$

where $\mathbf{r}=(X, Y, Z)$ is a vector of material coordinates. The rotation matrix is given by

$$
R_{s}=\left[\begin{array}{ccc}
\cos \left(n_{s} \theta\right) & -\sin \left(n_{s} \theta\right) & 0 \\
\sin \left(n_{s} \theta\right) & \cos \left(n_{s} \theta\right) & 0 \\
0 & 0 & 1
\end{array}\right]
$$

The number of the destination sector can be computed as

$$
n_{s}=\text { floor }\left[\operatorname{atan}\left(\frac{Y_{\text {destination }}}{X_{\text {destination }}}\right) / \theta\right]
$$

where the floor function returns the largest integer that is less than or equal to the argument.

## Reference

1. B. Lalanne and M. Touratier, "Aeroelastic Vibrations and Stability in Cyclic Symmetric Domains," The international journal of rotating machinery, vol. 6, no. 6, pp 445-452, 2000.

Model Library path: Structural_Mechanics_Module/Tutorial_Models/ impeller

## Modeling Instructions

## MODEL WIZARD

I Go to the Model Wizard window.
2 Click Next.
3 In the Add physics tree, select Structural Mechanics>Solid Mechanics (solid).
4 Click Next.

5 Find the Studies subsection. In the tree, select Preset Studies>Eigenfrequency.

## 6 Click Finish.

## GEOMETRY I

Import the pre-build geometry for the impeller from a file.

## Import I

I In the Model Builder window, right-click Model I>Geometry I and choose Import.
2 Go to the Settings window for Import.
3 Locate the Import section. Click the Browse button.
4 Browse to the model's Model Library folder and double-click the file impeller.mphbin.
5 Click the Import button.
6 Click the Go to Default 3D View button on the Graphics toolbar.
7 In the Model Builder window, right-click Geometry I and choose Build AlI.
8 Click the Go to Default 3D View button on the Graphics toolbar.
9 Click the Go to XY View button on the Graphics toolbar.
The complete geometry should look similar to that shown in Figure 1 and Figure 2.

## GLOBAL DEFINITIONS

## Parameters

I In the Model Builder window, right-click Global Definitions and choose Parameters.
2 Go to the Settings window for Parameters.
3 Locate the Parameters section. In the Parameters table, enter the following settings:

| NAME | EXPRESSION | DESCRIPTION |
| :--- | :--- | :--- |
| N | 8 | Number of sectors |
| theta | $2 * \mathrm{pi} / \mathrm{N}$ | Unit sector angle |
| m | 3 | Cyclic symmetry mode number |
| phi | $m *$ theta | Periodicity angle |
| p0 | $1 e 4[\mathrm{~Pa}]$ | Load magnitude |

## Variables I

I In the Model Builder window, right-click Global Definitions and choose Variables.
2 Go to the Settings window for Variables.

3 Locate the Variables section. In the Variables table, enter the following settings:

| NAME | EXPRESSION | DESCRIPTION |
| :--- | :--- | :--- |
| sector | floor $(\operatorname{atan2}(\mathrm{Y}, \mathrm{X}) /(2 * \mathrm{pi} / \mathrm{N}))$ | Sector selector |
| theta_sector | sector*2*pi/N | Sector angle |

## DEFINITIONS

Define a linear extrusion coupling to connect the boundaries of the sector, which is needed when setting up the Floquet periodicity conditions.

## Linear Extrusion I

I In the Model Builder window, right-click Model I>Definitions and choose Model Couplings>Linear Extrusion.
2 Go to the Settings window for Linear Extrusion.
3 Locate the Source Selection section. From the Geometric entity level list, choose Boundary.
4 Select Boundary 134 only.
5 Locate the Source Vertices section. Under Source vertex I, click Activate Selection.
6 Select Point 135 only.
7 Under Source vertex 2, click Activate Selection.
8 Select Point 141 only.
9 Under Source vertex 3, click Activate Selection.
10 Select Point 153 only.
II Under Source vertex 4, click Activate Selection.
I2 Select Point 176 only.
I3 Locate the Destination Vertices section. Under Destination vertex I, click Activate Selection.

14 Select Point 115 only.
15 Under Destination vertex 2, click Activate Selection.
16 Select Point 121 only.
17 Under Destination vertex 3, click Activate Selection.
18 Select Point 130 only.
19 Under Destination vertex 4, click Activate Selection.
20 Select Point 166 only.

21 Click to expand the Source section.
$\mathbf{Z}$ From the Source frame list, choose Material (X, Y, Z).
23 Click to expand the Destination section.
$\mathbf{2 4}$ From the Destination frame list, choose Material (X, Y, Z).
Add a general extrusion coupling to visualize the results computed for the sector over the full geometry.

## General Extrusion I

I In the Model Builder window, right-click Definitions and choose Model Couplings>General Extrusion.

2 Select Domain 8 only.
3 Go to the Settings window for General Extrusion.
4 Locate the Source section. From the Source frame list, choose Material (X, Y, Z).
5 Locate the Destination Map section. In the x-expression edit field, type $X * \cos (-$ theta_sector) $-Y * \sin (-$ theta_sector $)$.

6 In the $\mathbf{y}$-expression edit field, type $X * \sin (-$ theta_sector $)+Y * \cos (-$ theta_sector $)$.

7 In the $\mathbf{z}$-expression edit field, type $Z$.

## MODEL I

Add one more Solid Mechanics interface to use for the computations on the reduced geometry only.

## MODEL WIZARD

I In the Model Builder window, right-click Model I and choose Add Physics.
2 Go to the Model Wizard window.
3 In the Add physics tree, select Structural Mechanics>Solid Mechanics (solid).
4 Click Finish.
SOLID MECHANICS 2
I In the Model Builder window, click Model I>Solid Mechanics 2.
2 Select Domain 8 only.

## DEFINITIONS

## Variables 2

I In the Model Builder window, right-click Model I>Definitions and choose Variables.
2 Right-click Variables 2 and choose Rename.
3 Go to the Rename Variables dialog box and type Variables, postrocessing in the New name edit field.

4 Click OK.
5 Go to the Settings window for Variables.
6 Locate the Variables section. Click Load from File.
7 Browse to the model's Model Library folder and double-click the file impeller_variables.txt.

## Variables 3

I In the Model Builder window, right-click Definitions and choose Variables.
2 Right-click Variables 3 and choose Rename.
3 Go to the Rename Variables dialog box and type Variables, source in the New name edit field.

4 Click OK.
5 Go to the Settings window for Variables.
6 Locate the Geometric Entity Selection section. From the Geometric entity level list, choose Boundary.

7 Select Boundary 134 only.
8 Locate the Variables section. In the Variables table, enter the following settings:

| NAME | EXPRESSION |
| :--- | :--- |
| source_constr1 | $(u 2 * \cos ($ theta $)-v 2 * \sin (t h e t a))$ |
| source_constr2 | $(u 2 * \sin ($ theta $)+v 2 * \cos ($ theta) ) |
| source_constr3 | w2 |

Variables 4
I In the Model Builder window, right-click Definitions and choose Variables.
2 Right-click Variables 4 and choose Rename.
3 Go to the Rename Variables dialog box and type Variables, destination in the New name edit field.

4 Click OK.

5 Go to the Settings window for Variables.
6 Locate the Geometric Entity Selection section. From the Geometric entity level list, choose Boundary.

7 Select Boundary 112 only.

## Variables, destination

I In the Model Builder window, click Variables, destination.
2 Go to the Settings window for Variables.
3 Locate the Variables section. In the Variables table, enter the following settings:

| NAME | EXPRESSION |
| :--- | :--- |
| constr1 | linext1(source_constr1) |
| constr2 | linext1 (source_constr2) |
| constr3 | linext1 (source_constr3) |

## MATERIALS

I In the Model Builder window, right-click Model I>Materials and choose Open Material Browser.

2 Go to the Material Browser window.
3 Locate the Materials section. In the Materials tree, select Built-In>Aluminum.
4 Right-click and choose Add Material to Model from the menu.

## SOLID MECHANICS

## Fixed Constraint I

I In the Model Builder window, right-click Model I>Solid Mechanics and choose Fixed Constraint.

2 Select Boundaries $54,55,67,68,85,87,109$, and 113 only.
For a reduced geometry, you set up the Floquet periodicity conditions on the sector boundaries as pointwise constraints. To get access to this constraint setting, enable the advanced physics interface options.

3 In the Model Builder window's toolbar, click the Show button and select Advanced Physics Interface Options in the menu.

## SOLID MECHANICS 2

I In the Model Builder window, right-click 2 and choose Fixed Constraint.
2 Select Boundary 113 only.

## Pointwise Constraint I

I In the Model Builder window, right-click Solid Mechanics 2 and choose the boundary condition More>Pointwise Constraint.

2 Select Boundary 112 only.
3 Go to the Settings window for Pointwise Constraint.
4 Locate the Pointwise Constraint section. From the Constraint type list, choose User defined.
5 In the Constraint expression edit field, type constr1*exp(-i*phi)-u2.
6 In the Constraint force expression edit field, type test (constr1*exp(i*phi)-u2). Note that this setting imposes Hermitian symmetry on the constraint force Jacobian matrix in the discretized model.

## Pointwise Constraint 2

I Right-click Pointwise Constraint I and choose Duplicate.
2 Go to the Settings window for Pointwise Constraint.
3 Locate the Pointwise Constraint section. In the Constraint expression edit field, type constr2*exp(-i*phi)-v2.
4 In the Constraint force expression edit field, type test (constr2*exp(i*phi)-v2).

## Pointwise Constraint 3

I Right-click Pointwise Constraint 2 and choose Duplicate.
2 Go to the Settings window for Pointwise Constraint.
3 Locate the Pointwise Constraint section. In the Constraint expression edit field, type constr3*exp(-i*phi)-w2.
4 In the Constraint force expression edit field, type test (constr3*exp(i*phi)-w2).

## MESH I

I In the Model Builder window, click Model I>Mesh I.
2 Go to the Settings window for Mesh.
3 Locate the Mesh Settings section. From the Element size list, choose Fine.
4 Click the Build All button.
5 Click the Select Domains button on the Graphics toolbar.
6 Click the Go to Default 3D View button on the Graphics toolbar. The resulting mesh should look similar to that shown in Figure 3.

## STUDY I

Step I: Eigenfrequency
I In the Model Builder window, expand the Study I node, then click Step I: Eigenfrequency.

2 Go to the Settings window for Eigenfrequency.
3 Locate the Physics Selection section. In the table, enter the following settings:

| PHYSICS INTERFACE | USE |
| :--- | :--- |
| Solid Mechanics 2 (solid2) | $\times$ |

4 Locate the Study Settings section. In the Desired number of eigenfrequencies edit field, type 32.

## Solver I

I In the Model Builder window, right-click Study I and choose Show Default Solver.
2 Expand the Solver I node.
3 In the Model Builder window, expand the Dependent Variables I node, then click modl.u2.

4 Go to the Settings window for Field.
5 Locate the General section. Clear the Store in output check box.
6 In the Model Builder window, right-click Study I and choose Compute.

## RESULTS

## MODEL WIZARD

I In the Model Builder window, right-click the root node and choose Add Study.
2 Go to the Model Wizard window.
3 Find the Studies subsection. In the tree, select Preset Studies for Selected Physics>Eigenfrequency.

4 Find the Selected physics subsection. In the tree, select Solid Mechanics (solid2).
5 Click Finish.

STUDY 2
Step I: Eigenfrequency
I In the Model Builder window, click Study 2>Step I: Eigenfrequency.
2 Go to the Settings window for Eigenfrequency.

3 Locate the Physics Selection section. In the table, enter the following settings:

| PhYsics interface | USE |
| :--- | :--- |
| Solid Mechanics (solid) | $\times$ |

4 Locate the Study Settings section. In the Desired number of eigenfrequencies edit field, type 4.
To capture all possible eigenfrequencies, set up a sweep over the cyclic symmetry mode number $m$ in the range from 0 to $N / 2$, where $N$ is the total number of sectors.

## Parametric Sweep

I In the Model Builder window, right-click Study 2 and choose Parametric Sweep.
2 Go to the Settings window for Parametric Sweep.
3 Locate the Study Settings section. Under Parameter names, click Add.
4 Go to the Add dialog box.
5 In the Parameter names list, select $\mathbf{m}$ (Cyclic symmetry mode number).
6 Click the OK button.
7 Go to the Settings window for Parametric Sweep.
8 Locate the Study Settings section. Click the Range button.
9 Go to the Range dialog box.
10 In the Start edit field, type 0.
II In the Stop edit field, type N/2.
12 In the Step edit field, type 1.
13 Click the Add button.

## Solver 2

I In the Model Builder window, right-click Study 2 and choose Show Default Solver.
2 Expand the Solver 2 node.
3 In the Model Builder window, expand the Dependent Variables I node, then click modi.u.
4 Go to the Settings window for Field.
5 Locate the General section. Clear the Store in output check box.
6 In the Model Builder window, right-click Study 2 and choose Compute.
Note a nearly eight times reduction in the number of degrees of freedom, and thus of the memory required to compute the reduced model. However, the
computational time is approximately the same because you need to performe a sweep over all values of the periodicity parameter.

Modify the default plot of the eigenmode shape for the reduced model using the extrusion coupling variables to visualize the effective results over the full geometry.

## RESULTS

## Mode Shape (solid2)

I In the Model Builder window, expand the Mode Shape (solid2) node, then click Surface I.

2 Go to the Settings window for Surface.
3 Locate the Expression section. In the Expression edit field, type Disp2.
4 In the Model Builder window, expand the Surface I node, then click Deformation.
5 Go to the Settings window for Deformation.
6 Locate the Expression section. In the $\mathbf{X}$ component edit field, type U2.
7 In the $\mathbf{Y}$ component edit field, type V2.
8 In the $\mathbf{Z}$ component edit field, type W2.
9 Click the Plot button.

## Derived Values

Collect all the computed eigenfrequencies into tables.

## I In the Model Builder window, right-click Results>Derived Values and choose Global Evaluation.

2 Go to the Settings window for Global Evaluation.
3 Locate the Expression section. In the Expression edit field, type freq.

## 4 Right-click Global Evaluation I and choose Evaluate>New Table.

Note that the eigenfrequencies for the full geometry present groups of values very close to each other, eight frequencies in each group. This shows that vibrations of each of the eight blades of the impeller are only weakly coupled to the remaining structure, which happens because the central part has significantly larger effective bending stiffness compared to that of each blade. Hence, the eigenfrequencies in each group are close to the natural frequencies of a single blade (if computed assuming a fully fixed footing).

## I Right-click Derived Values and choose Global Evaluation.

2 Go to the Settings window for Global Evaluation.

3 Locate the Data section. From the Data set list, choose Solution 3.
4 From the Table rows list, choose Outer solutions.
5 Locate the Expression section. In the Expression edit field, type freq.
6 Right-click Global Evaluation 2 and choose Evaluate $>$ New Table.
Compare the values of the egenfrequencies computed by using the periodicity conditions to those found for the full geometry.

Next, add a load representing a periodic pressure perturbation in the stream, and thus on all the external boundaries of the impeller.

## SOLID MECHANICS

## Boundary Load I

I In the Model Builder window, right-click Model I>Solid Mechanics and choose Boundary Load.

2 Go to the Settings window for Boundary Load.
3 Locate the Boundary Selection section. From the Selection list, choose All boundaries.
4 Select Boundaries $1-3,5-18,20,21,23-53,56-66,69-75,77-84,88-107,110$, 111, 114-133, and 135-152 only.
You can do this by selecting all boundaries first, and then removing from the selection all the constraint boundaries and all the internal boundaries of the periodicity sectors.
5 Locate the Coordinate System Selection section. From the Coordinate system list, choose Boundary System I.

6 Locate the Force section. Specify the $\mathbf{F}_{\mathrm{A}}$ vector as

| 0 | $t 1$ |
| :--- | :--- |
| 0 | $t 2$ |
| $-p 0 * \exp (-j * m * \operatorname{atan} 2(Y, X))$ | $n$ |

SOLID MECHANICS 2
I In the Model Builder window, right-click 2 and choose Boundary Load.
2 Select Boundaries $114,115,119,123,125,130,131,133,136,137,142-146$, 150 , and 151 only.

3 Go to the Settings window for Boundary Load.

4 Locate the Coordinate System Selection section. From the Coordinate system list, choose Boundary System I.

5 Locate the Force section. Specify the $\mathbf{F}_{\mathrm{A}}$ vector as

| 0 | $t l$ |
| :--- | :--- |
| 0 | $t 2$ |
| $-p 0^{*} \exp (-j * m * \operatorname{atan} 2(Y, X))$ | $n$ |

Set up and perform the frequency-response analysis, first for the full model, and then for a sector of periodicity.

## MODEL WIZARD

I In the Model Builder window, right-click the root node and choose Add Study.
2 Go to the Model Wizard window.
3 Find the Studies subsection. In the tree, select Preset Studies for Selected Physics>Frequency Domain.

4 Click Finish.

STUDY 3
Step I: Frequency Domain
I In the Model Builder window, click Study 3>Step I: Frequency Domain.
2 Go to the Settings window for Frequency Domain.
3 Locate the Study Settings section. In the Frequencies edit field, type 200.
4 Locate the Physics Selection section. In the table, enter the following settings:

| PHYsics interface | USE |
| :--- | :--- |
| Solid Mechanics 2 (solid2) | $\times$ |

Switch off the generation of the default plot as that would be a plot of the von Mises stress, while you will be comparing the full and reduced structure responses in terms of displacements.

5 In the Model Builder window, click Study 3.
6 Go to the Settings window for Study.
7 Locate the Study Settings section. Clear the Generate default plots check box.

## Solver 9

I Right-click Study 3 and choose Show Default Solver.

2 In the Model Builder window, expand the Solver 9 node.
3 In the Model Builder window, expand the Dependent Variables I node, then click modi.u2.

4 Go to the Settings window for Field.
5 Locate the General section. Clear the Store in output check box.
6 In the Model Builder window, right-click Study 3 and choose Compute.

## RESULTS

3D Plot Group 3
I In the Model Builder window, right-click Results and choose 3D Plot Group.
2 Go to the Settings window for 3D Plot Group.
3 Locate the Data section. From the Data set list, choose Solution 4.
4 Right-click Results>3D Plot Group 3 and choose Rename.
5 Go to the Rename 3D Plot Group dialog box and type Diplacement (solid) in the New name edit field.

6 Click OK.

## Diplacement (solid)

I Right-click Results>3D Plot Group 3 and choose Surface.
2 In the Model Builder window, right-click Results>Diplacement (solid)>Surface I and choose Deformation.

3 Go to the Settings window for Deformation.
4 Locate the Scale section. Select the Scale factor check box.
5 In the associated edit field, type 25.
6 Click the Plot button.

## MODEL WIZARD

I In the Model Builder window, right-click the root node and choose Add Study.
2 Go to the Model Wizard window.
3 Find the Studies subsection. In the tree, select Preset Studies for Selected Physics>Frequency Domain.

4 Click Finish.

## STUDY 4

## Step I: Frequency Domain

I In the Model Builder window, click Study 4>Step I: Frequency Domain.
2 Go to the Settings window for Frequency Domain.
3 Locate the Physics Selection section. In the table, enter the following settings:

| PHYSICS INTERFACE | USE |
| :--- | :--- |
| Solid Mechanics (solid) | $\times$ |

4 Locate the Study Settings section. In the Frequencies edit field, type 200.
5 In the Model Builder window, click Study 4.
6 Go to the Settings window for Study.
7 Locate the Study Settings section. Clear the Generate default plots check box.

## Solver 10

I Right-click Study 4 and choose Show Default Solver.
2 In the Model Builder window, expand the Solver 10 node.
3 In the Model Builder window, expand the Dependent Variables I node, then click modi.u.

4 Go to the Settings window for Field.
5 Locate the General section. Clear the Store in output check box.
6 In the Model Builder window, right-click Study 4 and choose Compute.
For a frequency-response analysis, use of the reduced geometry gives significant gains in both the memory required and computational time needed.

## RESULTS

Set up a displacement plot for the reduced geometry and compare it to that for the full geometry.

## 3D Plot Group 4

I In the Model Builder window, right-click Results and choose 3D Plot Group.
2 Go to the Settings window for 3D Plot Group.
3 Locate the Data section. From the Data set list, choose Solution 5.
4 Right-click Results>3D Plot Group 4 and choose Rename.
5 Go to the Rename 3D Plot Group dialog box and type Displacement (solid2) in the New name edit field.

## 6 Click OK.

Displacement (solid2)
I Right-click Results>3D Plot Group 4 and choose Surface.
2 Go to the Settings window for Surface.
3 Locate the Expression section. In the Expression edit field, type Disp2.
4 Right-click Results>Displacement (solid2) $>$ Surface I and choose Deformation.
5 Go to the Settings window for Deformation.
6 Locate the Scale section. Select the Scale factor check box.
7 In the associated edit field, type 25.
8 Locate the Expression section. In the $\mathbf{X}$ component edit field, type U2.
9 In the $\mathbf{Y}$ component edit field, type V2.
10 In the $\mathbf{Z}$ component edit field, type W2.
II Click the Plot button.

